

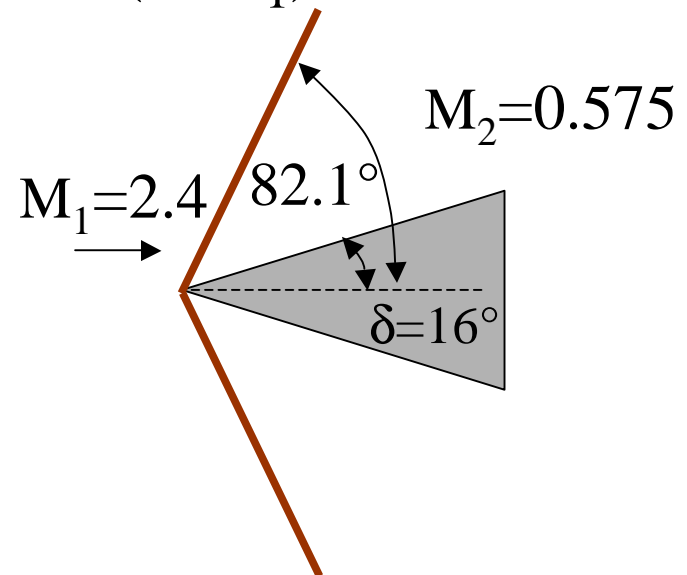
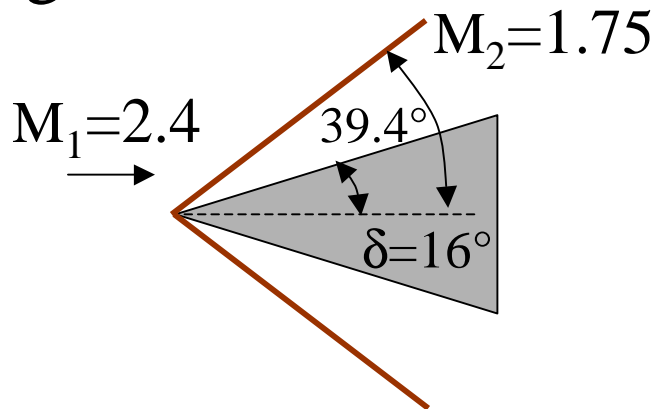
Strong and Weak Oblique Shocks

- As we have seen, it is possible to get two solutions to equation (VI.46)

$$\tan \delta = \left[\frac{2}{\tan \theta} (M_1^2 \sin^2 \theta - 1) \right] / \left[M_1^2 (\gamma + \cos 2\theta) + 2 \right]$$

– 2 possible values of θ for given (δ, M_1)

– e.g.,



- Examine graphical solution

Graphical Solution

- Weak shocks**

- smaller θ

- $\theta_{\min} = \mu$

- usually $M_2 > 1$

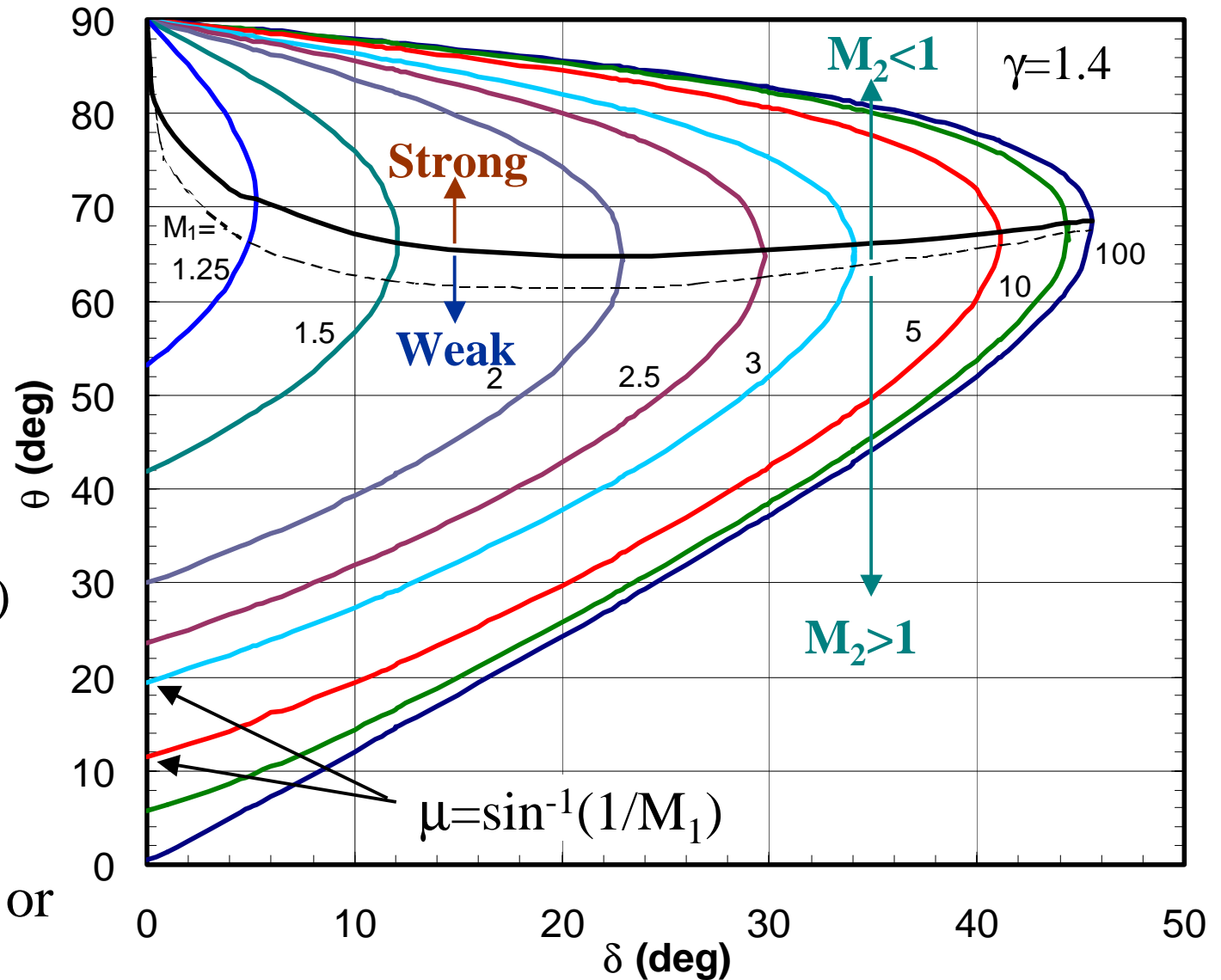
- Strong shocks**

- $\theta_{\max} = 90^\circ$
(normal shock)

- always $M_2 < 1$

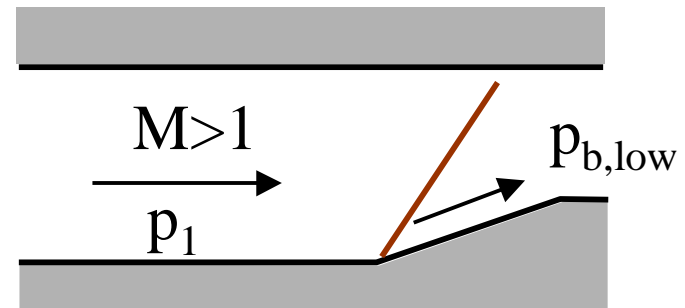
- Both for $\delta = 0$**

- no turn for normal shock or Mach wave

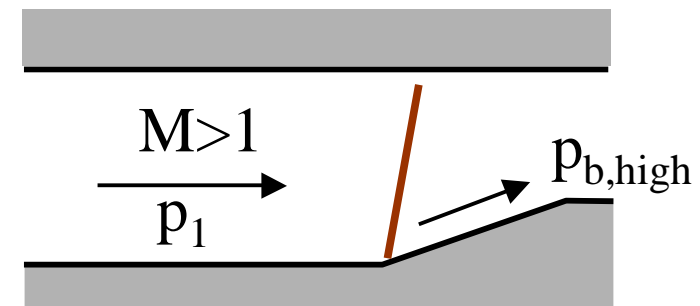


Which Solution Will Occur?

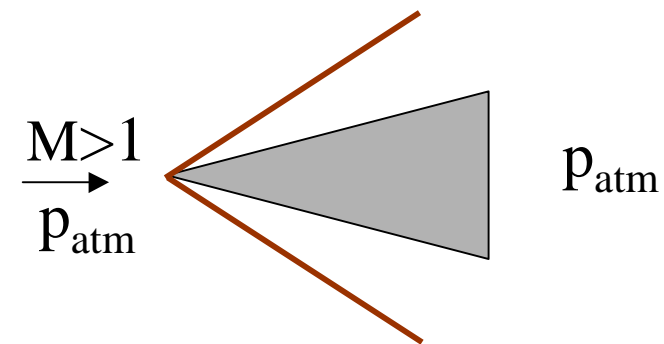
- Depends on upstream versus downstream pressure
 - e.g., back pressure
 - p_2/p_1 large for strong shock
small for weak shock



- Internal Flow
 - can have p_2 much higher or close to p_1

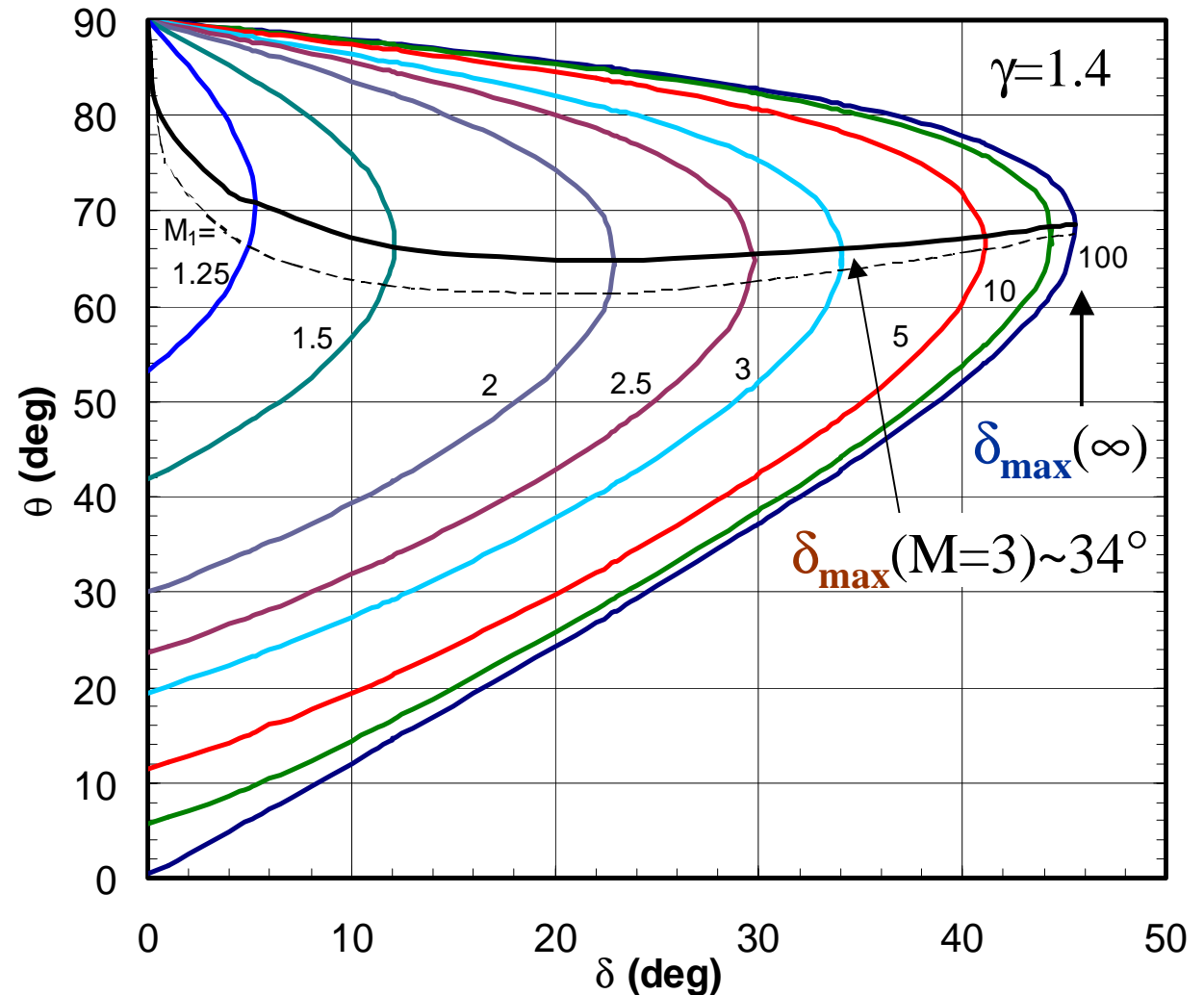


- External Flow
 - downstream pressure usually close to upstream p (both near p_{atm})



Maximum Turning Angle

- Given M_1 ,
no straight oblique shock solution for $\delta > \delta_{\max}(M)$
- Given δ ,
no solution for $M_1 < M_{1,\min}$
- Given fluid (γ),
no solution for any M_1 beyond δ_{\max}
e.g., $\sim 45.5^\circ$ ($\gamma=1.4$)



Detached Shock

- What does flow look like when no straight oblique shock solution exists?
 - **detached shock/bow shock**, sits ahead of body/turn
 - normal shock at centerline (flow subsonic to negotiate turn); curves away to weaker shock

