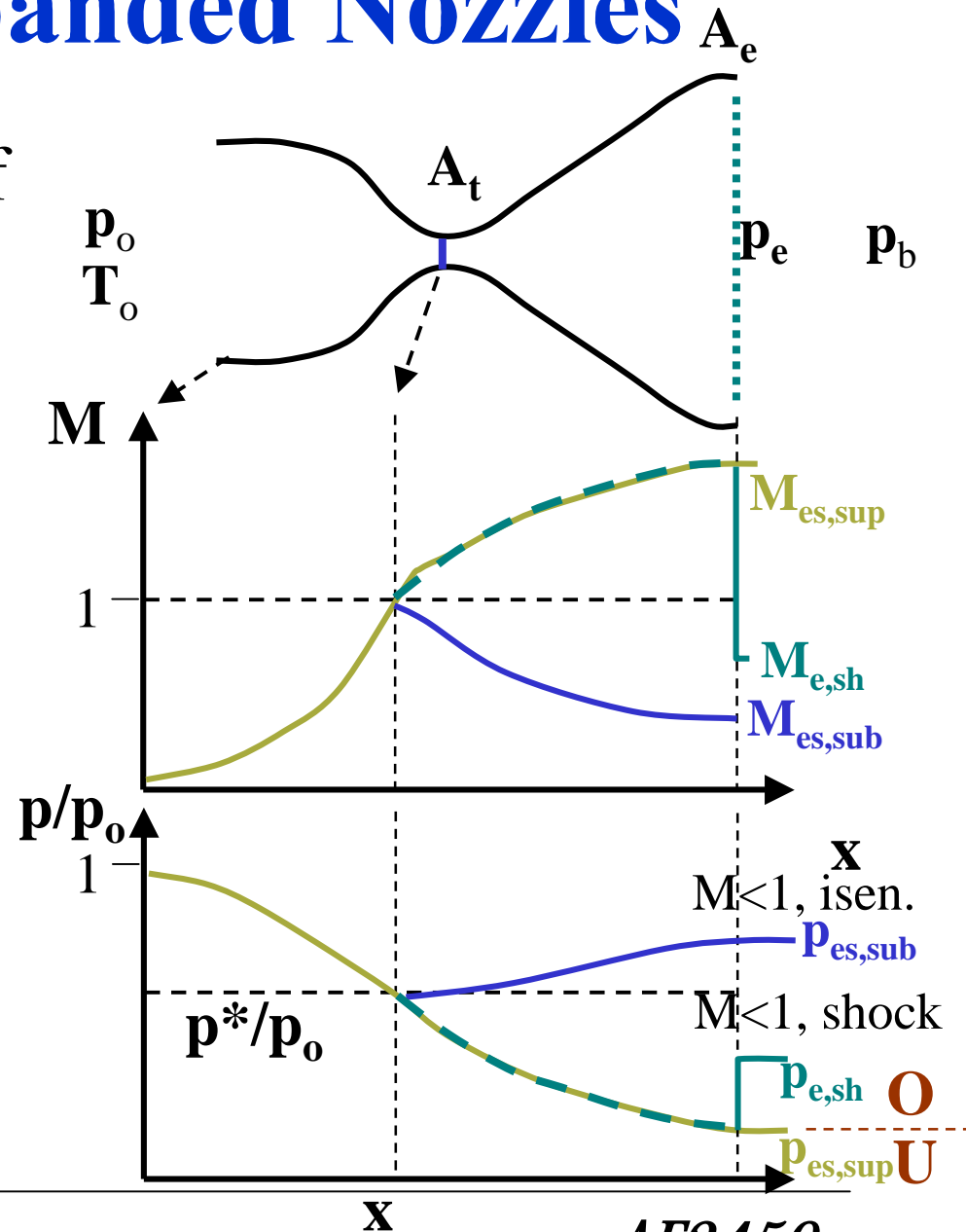


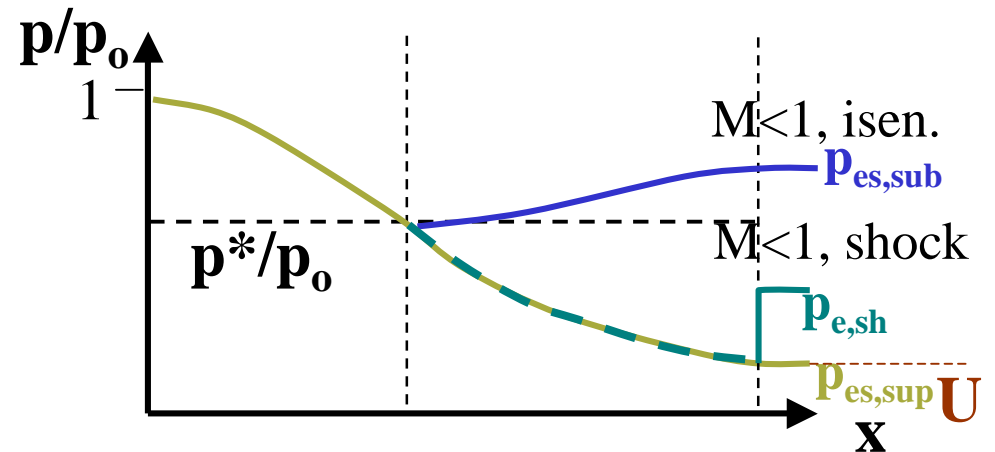
# Under/Overexpanded Nozzles

- Recall previous analysis of CD nozzles, e.g., as back pressure is reduced
- Have looked at range of  $p_b$  that produce
  - *isentropic* solutions or
  - *shocks in nozzle*
- What happens when
  - $p_b < p_{es,sup}$  (U)
  - $p_{es,sup} < p_b < p_{e,sh}$  (O)



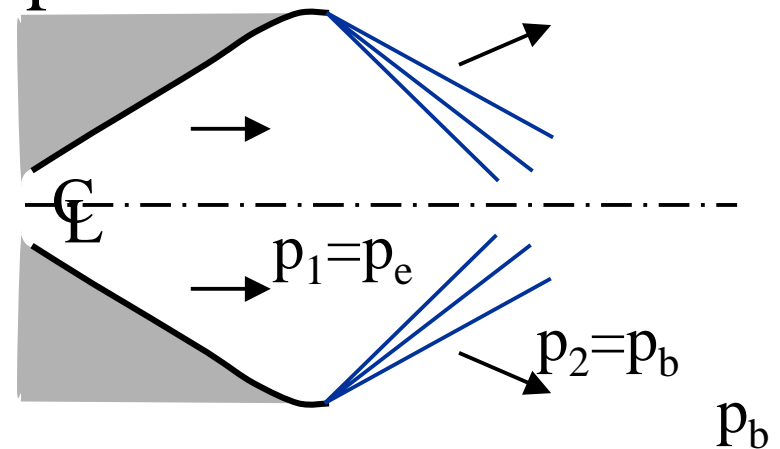
# Underexpanded Nozzles

- Start with  $p_b < p_{es,sup}$ 
  - **underexpanded case**
  - $p_e > p_b$  (not enough expansion has occurred)



- So **boundary condition** at exit requires a supersonic expansion process

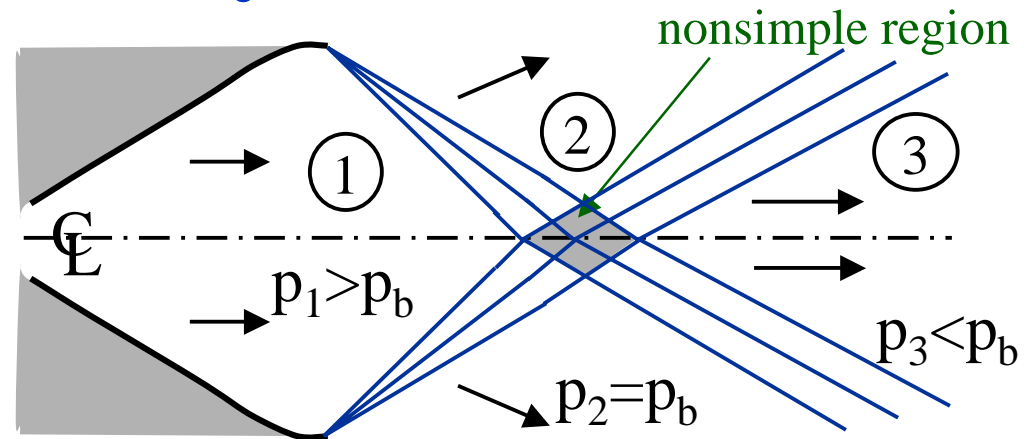
- ⇒ (2-d) Prandtl-Meyer flow
  - flow turns, pressure drops to back pressure
  - but flow must not cross centerline (symmetry line; like a wall)



# Centerline Boundary Condition

- Centerline

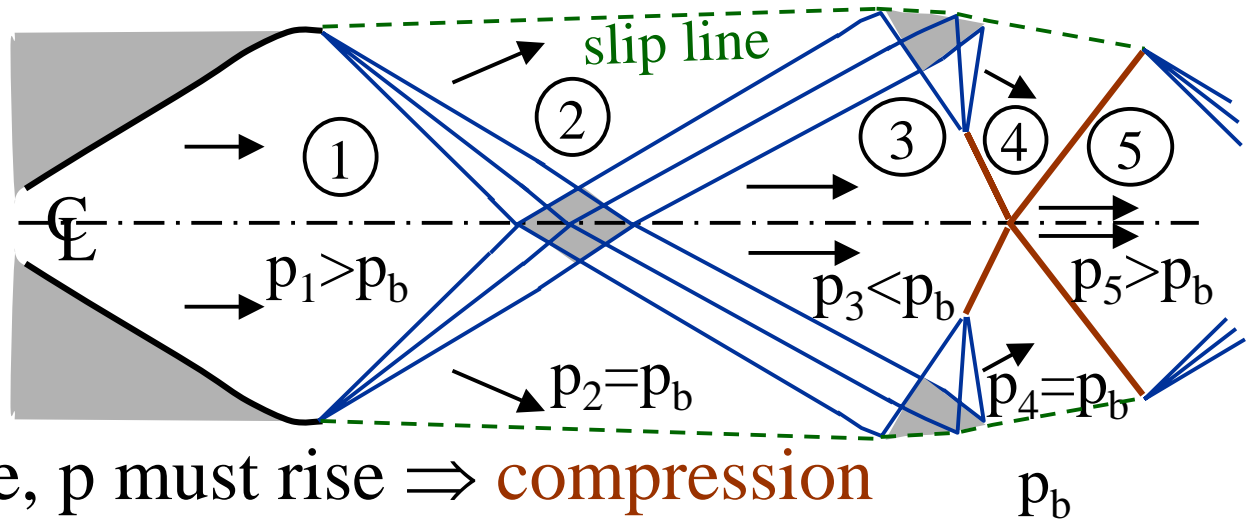
- flow along centerline must be parallel to centerline



- to get flow to turn inward again requires another PM expansion
- but second expansion means pressure in region 3 drops below back pressure ( $p_3 < p_2 = p_b$ )
- so now we will need to do something to match our pressure boundary condition at the edge of the jet

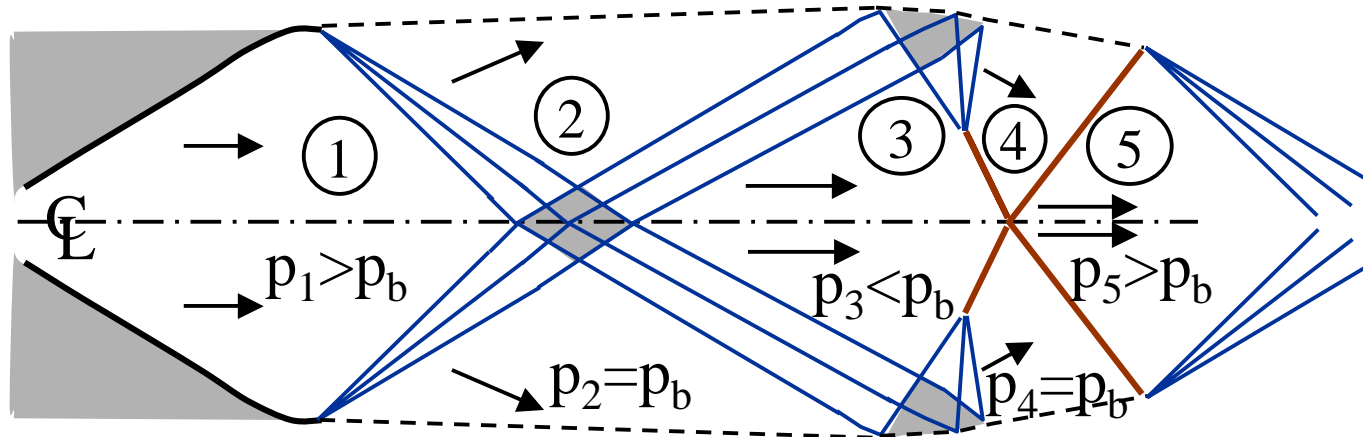
# Jet Boundary Condition

- Jet boundary
  - along “edge” of jet must satisfy pressure b.c.



- to match pressure,  $p$  must rise  $\Rightarrow$  **compression**
- can get PM compression waves
- combine to form oblique shocks, turn flow inward
- centerline (velocity) boundary condition requires reflected compression waves
- now pressure exceeds surrounding pressure ( $p_5 > p_b$ ), way we started, so process/cycle repeats itself

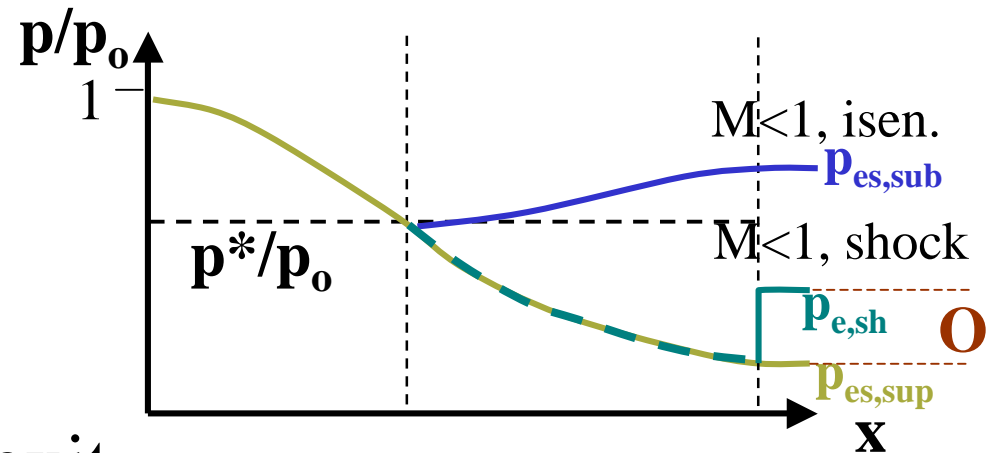
# Underexpanded Jet



- For  $p_e > p_b$ , nozzle produces a jet flow consisting of regions of *decreasing* and *increasing*  $p$  (or density)
  - inviscid flow (ideal), process would continue endlessly
  - viscous case (real), viscous losses and turbulent mixing with surroundings causes wave pattern to decay after small number of cycles

# Overexpanded Nozzles

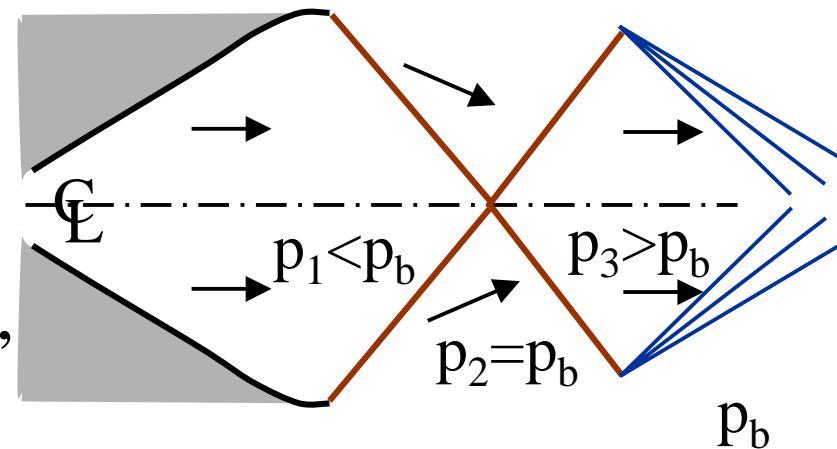
- $p_{es,sup} > p_b > p_{e,sh}$ 
  - **overexpanded case**
  - too much expansion has occurred



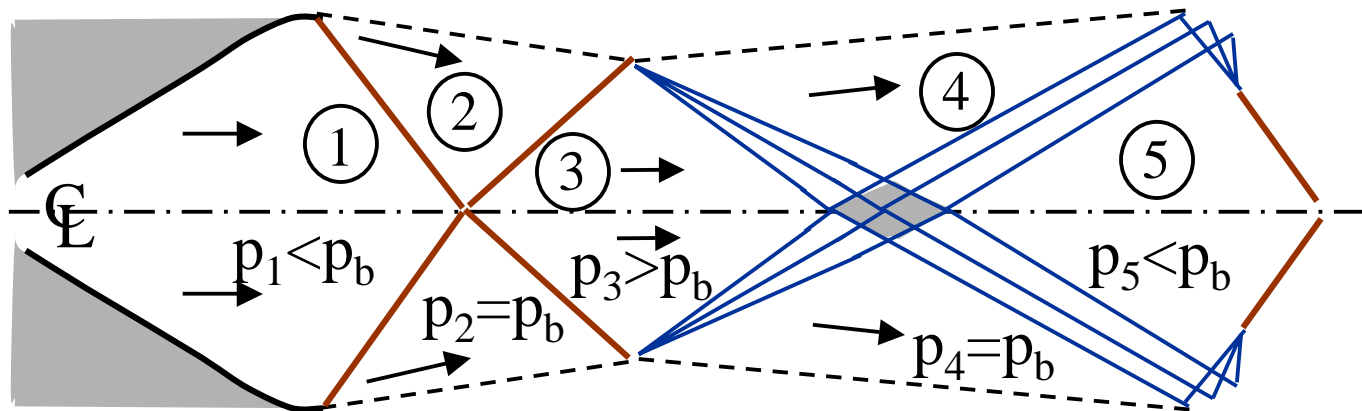
- So **boundary condition** at exit requires a supersonic compression process

⇒ oblique shock (2-d)

- flow turns, pressure rises to back pressure
- flow must not cross centerline, so get reflected shocks
- then expansion to match  $p$ , etc.



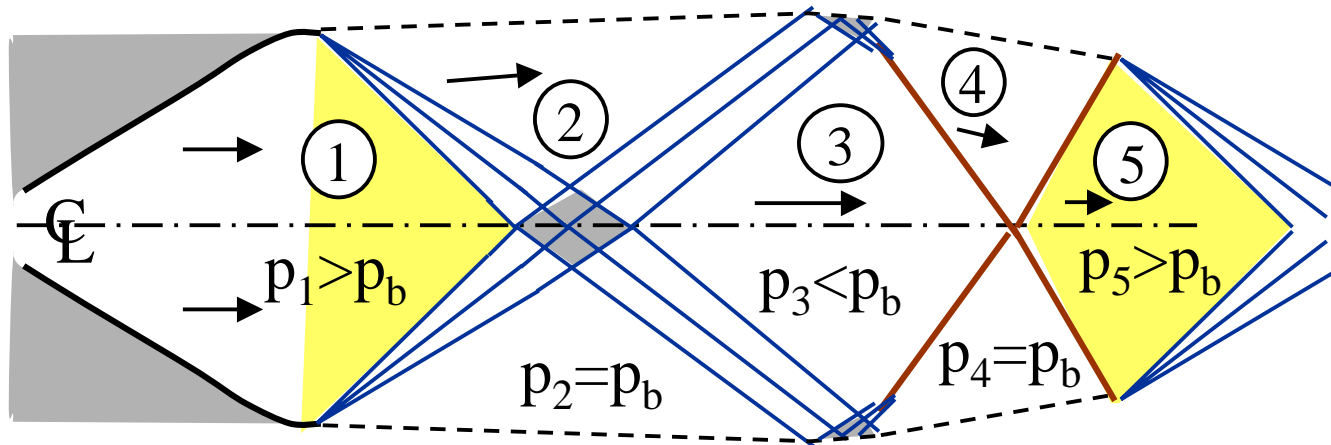
# Overexpanded Jet



- For  $p_e < p_b$ , nozzle produces jet flow consisting of regions of *increasing* and *decreasing* pressure (or density)
  - same pattern as for underexpanded case, just “out-of-phase” (compressions first, then expansions for overexpanded vs. expansions then compressions for underexpanded)

# Mach (Shock) Diamonds

- Regions of high pressure, also have high density (and temperature)
  - these hot, dense gases emit light (radiation)



SR-71 at takeoff