

Rocket Propulsion Basics

Under- and Over-Expanded Nozzles

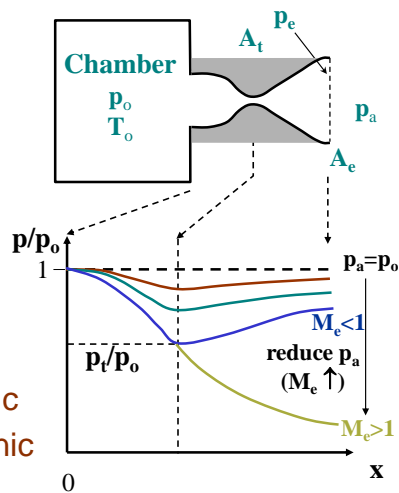
Expansion in CD Nozzles

- As ambient (back) pressure is reduced (or p_o increased), flow increases

- Once p_a/p_o low enough to get sonic flow at throat:

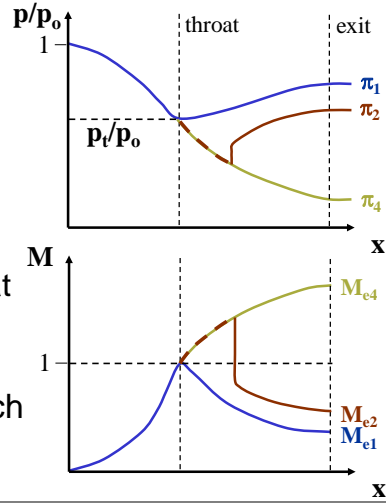
2 isentropic solutions

- higher $p_a(=p_e)$, subsonic
- lower $p_a(=p_e)$, supersonic



Nonisentropic Solutions

- What happens for p_d/p_o in between the isentropic solutions?
 - nonisentropic flow
- For $p_d/p_o < (p_d/p_o)_1 \equiv \pi_1$
 - flow starts to go supersonic after throat
- For $\pi_1 > \pi > \pi_4$, p must increase above supersonic isen. case to match $p_a \Rightarrow$ shock in diverging section

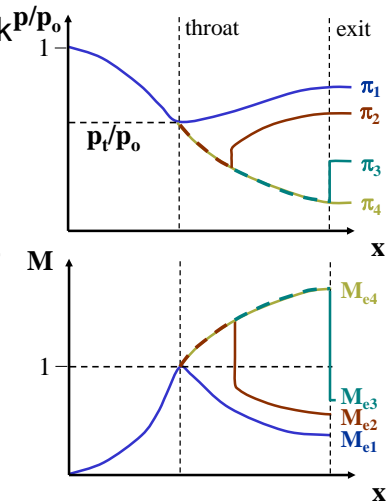


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Shocks Inside Nozzle

- Over what range of back pressures will there be shock in nozzle
 - until shock occurs at exit plane of nozzle
- So, question becomes - what is exit pressure when normal shock sits at exit?
 - combine isentropic flow with shock to match $p_e = p_a$

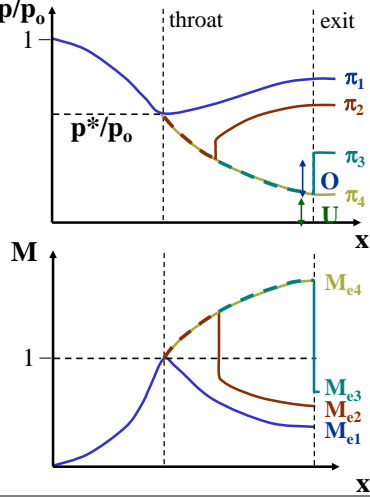


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Over- and Underexpanded Nozzles

- What happens if p_a/p_o goes below value where shock is at exit, $< \pi_3$
 - isentropic flow up to exit, **supersonic exhaust**
 - shocks (and expansions) **outside nozzle** (not normal shocks)
- $\pi_4 < \pi < \pi_3$: **Overexpanded**
- $\pi < \pi_4$: **Underexpanded**
- Generally, rocket nozzles over- or underexpanded*

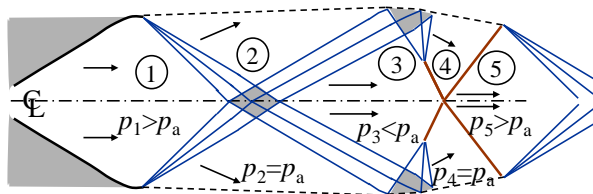


*except during startup/shutdown

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Underexpanded Nozzle Exhaust

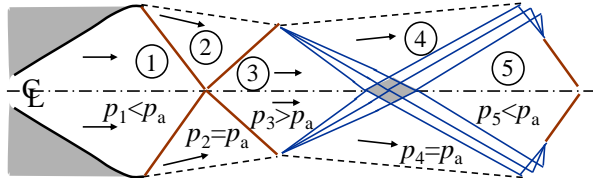


- For $p_e > p_a$, nozzle produces a jet flow consisting of regions of *decreasing* and *increasing* p (or density)
 - inviscid flow (ideal), process would continue endlessly
 - viscous case (real), viscous losses and turbulent mixing with surroundings causes wave pattern to decay after small number of cycles

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Overexpanded Nozzle Exhaust



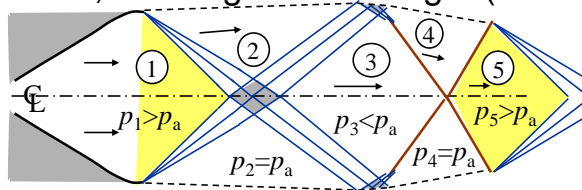
- For $p_e < p_a$, nozzle produces jet flow consisting of regions of *increasing* and *decreasing* pressure (or density)
 - same pattern as for underexpanded case, just “out-of-phase” (compressions first, then expansions for overexpanded vs. expansions then compressions for underexpanded)

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Mach (Shock) Diamonds

- Regions of high pressure, also have high density (and temperature)
 - these hot, dense gases emit light (radiation)



SR-71 at takeoff

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Altitude Variation in Exhaust Plume

- Which is high altitude test?



Credit: NASA



Credit: AEDC

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