



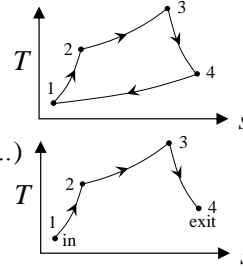
Thermodynamic Cycle Analysis

- One powerful use of TD is study of systems that manipulate energy (original motivation for TD)
 - **heat engines**: thermal energy conversion to work
 - convert ? energy → thermal energy → work (shaft)
 - **thermal transfer systems**: refriger., heat pumps
- These devices typically employ a working substance/fluid that goes through a (repeating) process

⇒ **thermodynamic cycle**

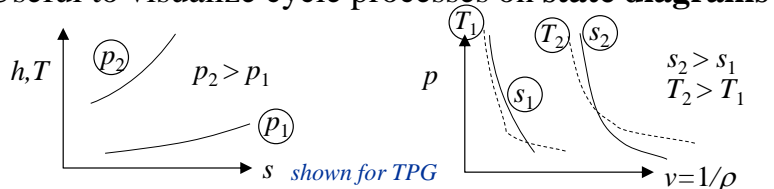
- **closed cycle** (*steam power plant, refriger.,...*)
 - same working fluid continuously circulates through cycle
- **open cycle** (*jet engine, rocket, car engine,...*)
 - fluid enters and leaves device, but new fluid (at same initial condition) keeps replacing exhausted fluid

but same analysis used for both



Ideal Cycles and State Diagrams

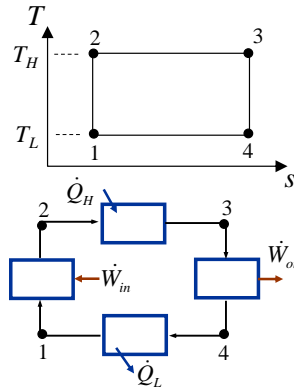
- Simply analysis of many real devices with ideal cycle
 1. ideal fluid, e.g., thermally, calorically perf. gas
 2. simplify processes, e.g., combustor replaced by nonreacting heat exchanger
 3. open → closed, e.g., surroundings (atmos.) = ht xchngr
 4. reversible, i.e., all components internally reversible
- Useful to visualize cycle processes on **state diagrams**





Carnot Cycle

- Start by examining a completely reversible cycle
- Carnot cycle:** 4 processes
 - 2 rev. isentropic (1→2, 3→4)
 - work, but no heat transfer
 - 2 rev. isothermal (2→3, 4→1)
 - heat transfer (and work?)
- What is efficiency of a device that follows this cycle?
- For heat engine *cycle or thermal efficiency*



(II.18)

$$\eta = \frac{\text{net work out}}{\text{energy added as heat}}$$

← what we want
← what we "pay for"

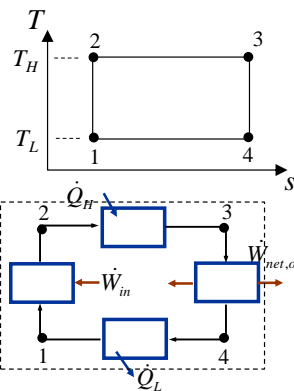


Carnot Cycle

- $\eta = \frac{\text{net work out}}{\text{energy added as heat}}$
- To find η use energy conservation
 - consider C.S. around engine

$$\dot{W}_{net,out} = \dot{Q}_H - \dot{Q}_L$$
 - so $\eta = \frac{\dot{Q}_H - \dot{Q}_L}{\dot{Q}_H} = 1 - \frac{Q_L}{Q_H}$
 - if rev. $\Delta S = Q/T$

$$= 1 - \frac{T_L(s_4 - s_1)}{T_H(s_3 - s_2)}$$



T_L	T_H	$\eta\%$
300	600	50.0
300	2400	87.5
200	2400	91.7

$$\eta_{Carnot} = 1 - \frac{T_L}{T_H}$$

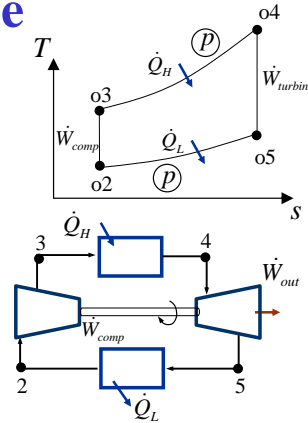
(II.19)

- get max η for $T_H \gg T_L$; add heat at high T as possible (reject heat at low T as possible)
- even though every process in Carnot cycle is rev., can't get 100% effic. (unless $T_L=0$)
- can show no 2 T heat engine has higher η than Carnot



Brayton Cycle

- Ideal cycle for jet engines, gas turbines
 - 2 rev+adiabatic processes
 - compressor 2→3
 - turbine (expansion) 4→5
 - 2 rev+isobaric processes
 - combustor (heat add) 3→4
 - heat exchanger 5→2



Efficiency

$$\eta = \frac{\dot{W}_{turb} - \dot{W}_{comp}}{\dot{Q}_H} = \frac{\dot{Q}_H - \dot{Q}_L}{\dot{Q}_H}$$

$$= \frac{\dot{m}[(h_{04} - h_{05}) - (h_{03} - h_{02})]}{\dot{m}(h_{04} - h_{03})} = 1 - \frac{T_{05} - T_{02}}{T_{04} - T_{03}}$$

tpg, cpg $\eta \uparrow$ for high $T_{04} - T_{03}$ (material limit) and low $T_{05} - T_{02}$ (reduces wasted "heat")

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Brayton Cycle

- Efficiency

tpg, cpg

$$\eta = 1 - \frac{T_{05} - T_{02}}{T_{04} - T_{03}}$$

$$= 1 - \frac{\frac{1}{T_{03}} \frac{T_{05}}{T_{04}} - \frac{1}{T_{04}} \frac{T_{02}}{T_{03}}}{\frac{T_{04}}{T_{03}} - \frac{T_{03}}{T_{04}}}$$

$$\frac{T_{02}}{T_{03}} = \left(\frac{p_{02}}{p_{03}}\right)^{\gamma-1/\gamma} = \frac{T_{05}}{T_{04}}$$

$$\Pr \equiv p_{03}/p_{02}$$

cycle pressure ratio

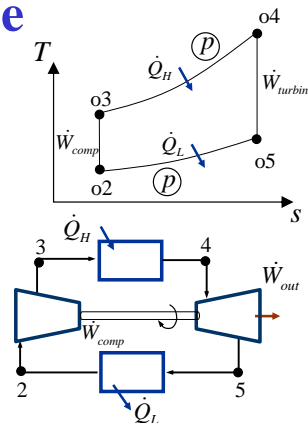
$$= 1 - \frac{\left(\frac{1}{T_{03}} - \frac{1}{T_{04}}\right) \left(\frac{1}{\Pr}\right)^{\gamma-1/\gamma}}{\frac{1}{T_{03}} - \frac{1}{T_{04}}}$$

Pr	$\eta\%$ ($\gamma=1.4$)
2	18
10	48
30	62

$$\eta_{Brayton} = 1 - \left(\frac{1}{\Pr}\right)^{\gamma-1/\gamma}$$

(II.20)

$\eta_{Brayton} \uparrow$ for high pressure ratio
– not strictly function of temperature,
but Pr limited by max T allowed



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