



Ideal Gas Mixtures

- Most propulsion applications do not involve one “pure” gas – involve gas mixtures
 - for example: air (O_2 , N_2 ...), engine exhaust (products of combustion)
- How to calculate properties of ideal gas mixture?
 - can use mixture averaged properties, $p = \rho R_{mix} T$ $R_{mix} = \frac{R}{MW_{mix}}$; $MW_{mix} = \sum_i \chi_i MW_i$ (II.13)
 - or sum up $p = \sum_i p_i$ $p_i = \rho_i R_i T$ $p_i = \chi_i p$

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Calculating Properties of Mixtures

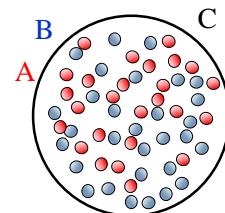
- What about other properties?
 - internal energy E or enthalpy H
 - e.g., gas C composed of part A and B

mixture averaged properties

$$\Delta H_{C_{1,2}} = m_{mix} \int_{T_1}^{T_2} c_{p_{mix}} dT$$

$$c_{p_{mix}} = \sum_i Y_i c_{p_i} \quad Y_i = m_i / m_{mix}$$

or



simple summation

$$\Delta H_{C_{1,2}} = \Delta H_{A_{1,2}} + \Delta H_{B_{1,2}}$$

$$\Delta H_{C_{1,2}} = m_A \{h_A(T_2) - h_A(T_1)\} + m_B \{h_B(T_2) - h_B(T_1)\}$$

$$\Delta H_{C_{1,2}} = m_A \int_{T_1}^{T_2} c_{p_A} dT + m_B \int_{T_1}^{T_2} c_{p_B} dT$$

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Ideal Gas Mixture Entropy

- What about entropy S ?

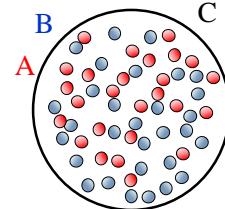
– use mixture averaged properties

$$\Delta S_{C_{1,2}} = m_{mix} \left[\int_{T_1}^{T_2} c_{p_{mix}} \frac{dT}{T} - R_{mix} \ln \left(\frac{p_2}{p_1} \right) \right]$$

– or sum up components

$$S_C(T, p) = S_A(T, ?) + S_B(T, ?)$$

all components have same T , but what pressure should we use for each component? ... **their partial pressure**



$$\Delta S_{C_{1,2}} = m_A \left\{ \int_{T_1}^{T_2} c_{p_A} \frac{dT}{T} - R_A \ln \left(\frac{p_{2A}}{p_{1A}} \right) \right\} + m_B \left\{ \int_{T_1}^{T_2} c_{p_B} \frac{dT}{T} - R_B \ln \left(\frac{p_{2B}}{p_{1B}} \right) \right\}$$

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Example

- **Given:** air at 1 atm and 300 K compressed to 10 atm and 700 K
- **Find:** change in entropy per unit mass
- **Assume:** synthetic air is 79% N₂ and 21% O₂ (by mole); N₂, O₂ are TPG and CPG under these conditions
- **Analysis:**



1) mix. avg properties

$$\Delta S_{12} = c_{p_{air}} \ln \left(\frac{T_2}{T_1} \right) - R_{air} \ln \left(\frac{p_2}{p_1} \right)$$

$$MW_{air} = 0.79(28 \text{ kg/kmol}) + 0.21(32 \text{ kg/kmol}) = 28.85 \text{ kg/kmol}$$

$$R_{air} = \frac{\bar{R}}{MW_{air}} = \frac{8314 \text{ J/kmolK}}{28.85 \text{ kg/kmol}} = 288 \text{ J/kgK}$$

$$c_{p_{air}} \approx \frac{7}{2} R_{air} = 1.01 \text{ kJ/kgK}$$

$$= 1.01 \frac{\text{kJ}}{\text{kgK}} \ln \left(\frac{7}{3} \right) - 0.288 \frac{\text{kJ}}{\text{kgK}} \ln \left(\frac{10}{1} \right)$$

$$= 0.191 \frac{\text{kJ}}{\text{kgK}}$$

S increased...why?

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Example

- **Analysis:**

2) summation

$$\Delta S_{12,air} = \Delta S_{12,N_2} + \Delta S_{12,O_2}$$

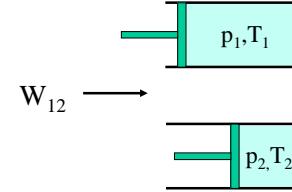
$$m_{air}\Delta s_{12,air} = m_{N_2}\Delta s_{12,N_2} + m_{O_2}\Delta s_{12,O_2}$$

$$\Delta s_{12,air} = Y_{N_2}\Delta s_{12,N_2} + Y_{O_2}\Delta s_{12,O_2}$$

$$= 0.767 \frac{kg_{N_2}}{kg_{air}} \left(0.1968 \frac{kJ}{kg_{N_2} K} \right)$$

$$+ 0.233 \frac{kg_{O_2}}{kg_{air}} \left(0.1722 \frac{kJ}{kg_{O_2} K} \right)$$

$$= 0.191 \frac{kJ}{kgK}$$



$$Y_{N_2} = \chi_{N_2} \frac{MW_{N_2}}{MW_{air}} = 0.79 \frac{28.01}{28.8} = 0.767$$

$$Y_{O_2} = 1 - Y_{N_2} = 0.233$$

$$\Delta s_{12,N_2} = \frac{8314}{28.01} \frac{J}{kg_{N_2} K} \left\{ \frac{7}{2} \ln\left(\frac{7}{3}\right) - \ln\left(\frac{7.9}{0.79}\right) \right\}$$

$$= 0.1968 kJ/kg_{N_2} K$$

$$\Delta s_{12,O_2} = \frac{8314}{32.0} \frac{J}{kg_{O_2} K} \left\{ \frac{7}{2} \ln\left(\frac{7}{3}\right) - \ln\left(\frac{2.1}{0.21}\right) \right\}$$

$$= 0.1722 kJ/kg_{O_2} K$$

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