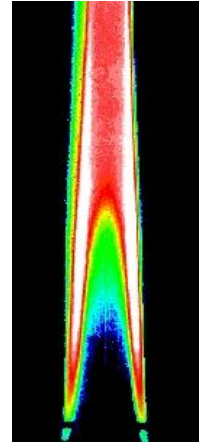
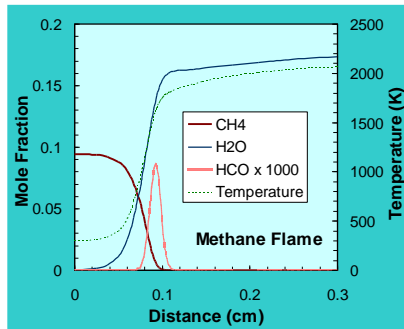


Introduction to Turbulent Combustion: Turbulent Premixed Flames

Jerry Seitzman



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Overview

- The specific goals of this section:
 1. Refine combustion regime paradigm for turbulent premixed systems by comparing to laminar flame scales
 - Borghi-Peters diagram
 2. Introduce turbulent flame speed S_T
 3. Examine simple models for S_T scaling (e.g., vs. S_L)
 4. Comments on local extinction and turbulent combustion modeling based on G equation

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Length and Velocity Ratios

- Compare turbulent to laminar premixed scales
- **Length scale ratio**
 - start by comparing turbulent macroscale to a characteristic premixed flame dimension
 - laminar (unstretched) flame thickness

$$\frac{\ell_o}{\delta_{fL}} \approx \frac{\ell_o}{\alpha/S_L^o} \approx \frac{\ell_o}{\sqrt{\alpha\tau_{chem}}} \approx \frac{\ell_o}{\sqrt{\nu\tau_{chem}}} \quad \text{assuming } Pr \sim 1, \text{ so } \nu \sim \alpha$$

$$\frac{\ell_o}{\delta_{fL}} \approx \left(\frac{\ell_o/\sqrt{2q}}{\tau_{chem}} \frac{\ell_o\sqrt{2q}}{\nu} \right)^{1/2}$$

$$\frac{\ell_o}{\delta_{fL}} \approx (Da_t Re_{\ell_o})^{1/2} \quad (X.4)$$

Length and Velocity Ratios

- **Velocity ratios**
 - now compare rms turbulent velocity fluctuation to laminar flame speed

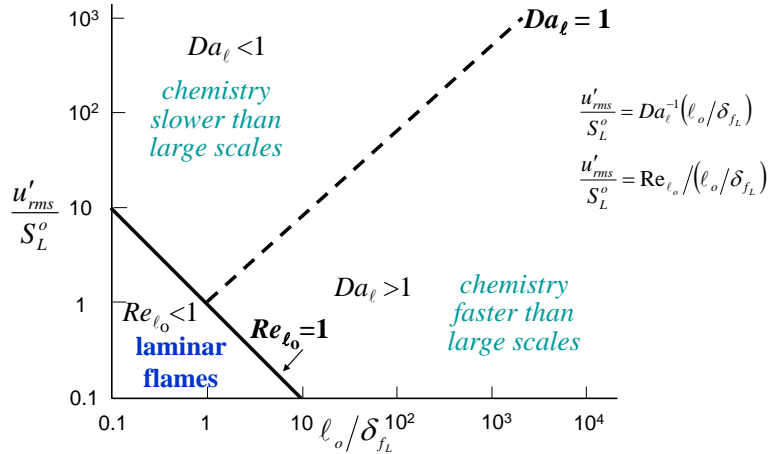
$$\frac{u'_{rms}}{S_L^o} = \frac{\sqrt{2q}}{S_L^o} \approx \frac{\sqrt{2q}}{\sqrt{\nu/\tau_{chem}}} \quad \text{again for } Pr \sim 1$$

$$\approx \left(\frac{\sqrt{2q}}{\nu} \sqrt{2q\tau_{chem}} \right)^{1/2} = \left(\frac{\sqrt{2q}\ell_o}{\nu} \frac{\sqrt{2q}}{\ell_o} \tau_{chem} \right)^{1/2}$$

$$\left. \begin{array}{l} \frac{u'_{rms}}{S_L^o} \approx (Re_{\ell_o}/Da_t)^{1/2} \\ \text{recall (X.4)} \quad \frac{\ell_o}{\delta_{fL}} \approx (Da_t Re_{\ell_o})^{1/2} \end{array} \right\} \Rightarrow \begin{array}{l} \boxed{\frac{u'_{rms}}{S_L^o} = Da_t^{-1}(\ell_o/\delta_{fL})} \quad (X.5) \\ \boxed{\frac{u'_{rms}}{S_L^o} = Re_{\ell_o}/(\ell_o/\delta_{fL})} \quad (X.6) \end{array}$$

Length and Velocity Ratios

- Graphically representing these relationships



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Small Scales and Karlovitz Number

- Above length and velocity ratios are for large scale turbulent structures
- What about small scales, i.e., ℓ_k vs. δ_{fL} ?
 - previously saw $\ell_k = \ell_o Re_{\ell_o}^{-3/4} \Rightarrow \ell_k/\delta_{fL} = (\ell_o/\delta_{fL}) Re_{\ell_o}^{-3/4}$
 - using (X.4) $\ell_k/\delta_{fL} = (Da_\ell Re_{\ell_o})^{1/2} Re_{\ell_o}^{-3/4}$
 $\ell_o/\delta_{fL} \approx (Da_\ell Re_{\ell_o})^{1/2} = (Da_\ell Re_{\ell_o}^{-1/2})^{1/2}$
 - but from (X.3) $Da_\ell \approx Re_{\ell_o}^{1/2} Da_k \Rightarrow \ell_k/\delta_{fL} \approx Da_k^{1/2} = \left(\frac{\tau_k}{\delta_{fL}/S_L^o} \right)^{1/2} \frac{\text{Kolmogorov eddy time}}{\text{laminar flame transit time}}$
 - flame turbulent Karlovitz #**
 $\equiv 1/Ka_L$

$$\ell_k/\delta_{fL} \approx Ka_L^{-1/2} \quad (X.7)$$

@ $Ka_L=1 \Rightarrow \ell_k=\delta_{fL}$
 $Ka_L>1 \Rightarrow \ell_k<\delta_{fL}$

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Karlovitz Number (con't)

- Velocity ratio?

– from (X.5) and (X.3)

$$\frac{u'_{rms}}{S_L} = Da_k^{-1} \frac{\ell_o}{\delta_{fL}} \quad Da_k \approx Re_{\ell_o}^{1/2} Da_k \quad \frac{u'_{rms}}{S_L} \approx (Da_k^{-1} Re_{\ell_o}^{-1/2}) \frac{\ell_o}{\delta_{fL}}$$

$$\approx Ka_L Re_{\ell_o}^{-1/2} \frac{\ell_o}{\delta_{fL}}$$

– from (X.6)

$$Re_{\ell_o} = (u'_{rms}/S_L) (\ell_o/\delta_{fL}) \quad \frac{u'_{rms}}{S_L} \approx Ka_L \left[\frac{u'_{rms}}{S_L} \frac{\ell_o}{\delta_{fL}} \right]^{-1/2} \frac{\ell_o}{\delta_{fL}}$$

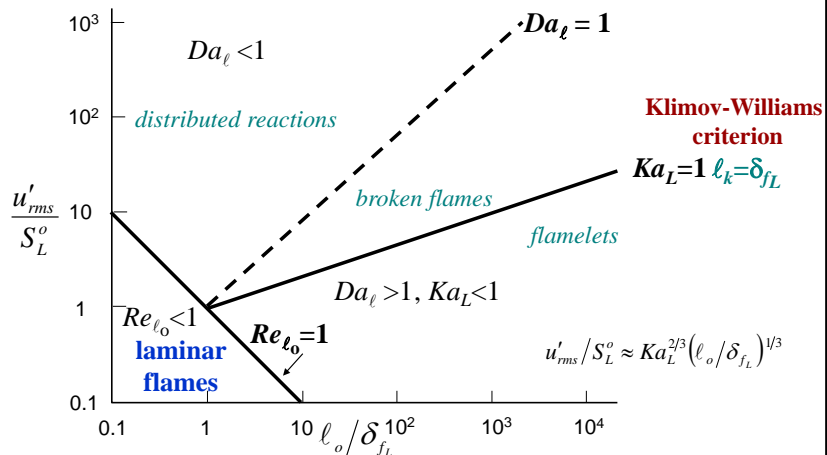
$$(u'_{rms}/S_L)^{3/2} \approx Ka_L (\ell_o/\delta_{fL})^{1/2}$$

Ka_L scaling for velocity and length ratios

$$u'_{rms}/S_L \approx Ka_L^{2/3} (\ell_o/\delta_{fL})^{1/3} \quad (X.8)$$

Karlovitz Scaling

- Add these relations to graph

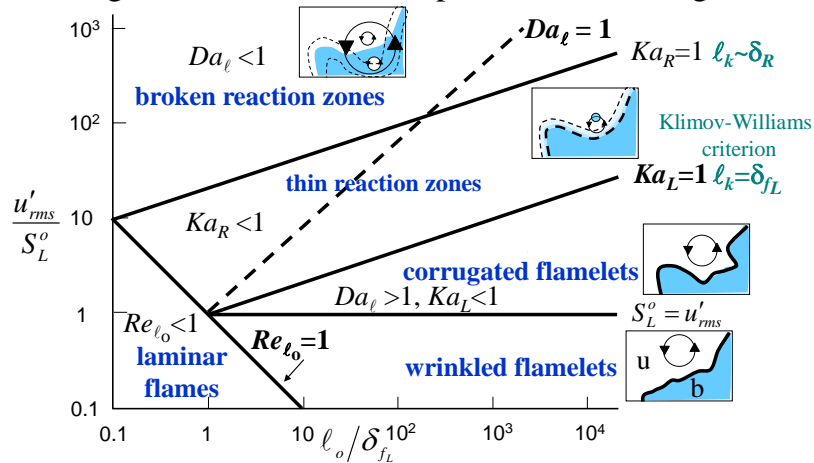


Reaction Zone Ka

- Can define a second Karlovitz number based on the “reaction zone” thickness δ_R
 - instead of the larger flame thickness δ_{fL}
 - recall $\delta_R / \delta_{fL} \sim 1/Ze$, Zelodvich number
- So with (X.7)
$$Ka_L \approx (\delta_{fL} / \ell_k)^2 \quad \frac{Ka_R}{Ka_L} = (\delta_R / \delta_{fL})^2 = Ze^{-2} \quad (X.9)$$
- Recall for many **room temperature**, HC-air flames, $Ze \sim 10$
 - $Ka_R=1$ when $Ka_L \sim 100$

Borghi-Peters Diagram

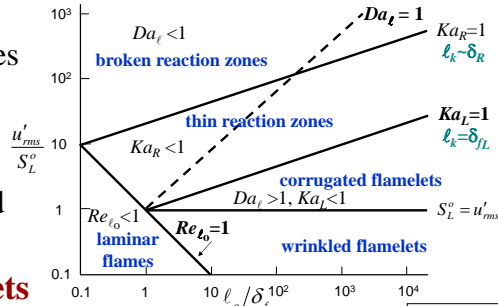
- Scalings used to differentiate premixed flame regimes



Premixed Flame Regimes

Flamelets

- all turbulent scales larger than flame thickness
- turbulent flame acts like distorted laminar flame



Wrinkled Flamelets

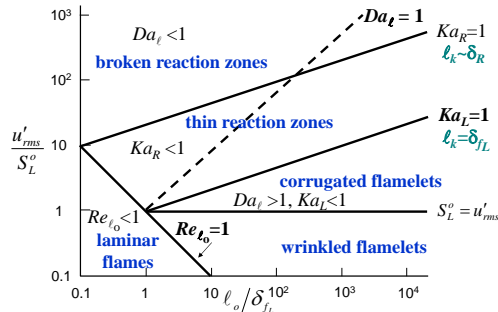
- $u'_{rms} < S_L^o$
- laminar flame propagation dominates turbulent displacement, i.e., any corrugations in flame front being induced by turbulence*



Premixed Flame Regimes

Corrugated Flamelets

- $u'_{rms} > S_L^o > u'_k$
- large eddies push flame, cause significant corrugation
- small eddies too slow and large to impact (internal) flame structure
- flame structure remains same as quasi-steady (but stretched) laminar flame*

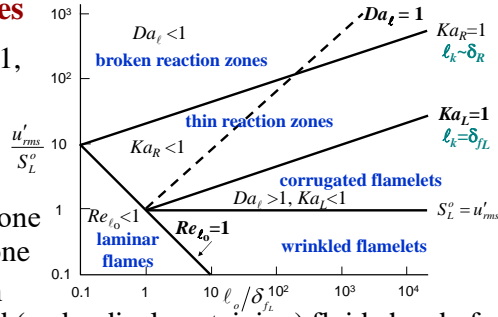


Premixed Flame Regimes



- **Thin Reaction Zones**

- $Da_{\ell_o} \sim O(1), Ka_L > 1, Ka_R < 1$
- $\delta_R < \ell_k < \delta_{fL}$
- small scales distort "preheat" zone but not reaction zone
- "small" eddies can transport preheated (and radical containing) fluid ahead of laminar diffusion length
- these eddies can enhance flame propagation or lead to local extinctions



- but can withstand instantaneous $Ka > Ka_{steady, extinct}$

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Turbulent Flame Speed

- What is **avg** propagation speed of turbulent flame, S_T
- Use analogy to laminar flame speed, S_L
 - example, consider **flamelet regime**
 - instantaneous, local propagation by S_L
 - for avg., let's define based on reactant consumption rate

$$\dot{m} = \rho_u S_L A$$

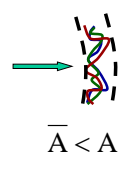
$$\dot{m} = \rho_u S_T \bar{A} \quad t=1$$

avg

$$S_T \equiv \dot{m} / \rho_u \bar{A} \quad t=2$$

(X.10)

t=3



interpretation: reactant consumption enhanced by turbulence due to flame area increase

$$\bar{A} < A_{inst} \Rightarrow S_T > S_L$$

$$S_T = S_L(\phi, T, p, \dots + \bar{u}, u', e(k))$$

mixture props. flow/turb. properties

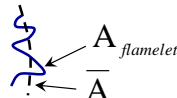
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S_T for Flamelet Regime

- If instantaneous propagation same as laminar flamelets

$$\frac{S_T}{S_L} = \frac{A_{\text{flamelets}}}{A}$$


- S_L not necessarily constant
 - local flow properties may vary, e.g. T
 - turbulence leads to stretch, $S_L \neq S_L^0$
- Simplest model: no S_L change (flame only “pushed”)
- Danköehler suggested wrinkling related to u'_{rms}

$$\frac{A_{\text{flamelets}}}{A} \sim 1 + \frac{u'_{rms}}{S_L} \Rightarrow \frac{S_T}{S_L} \cong 1 + \frac{u'_{rms}}{S_L} \Rightarrow S_T/S_L = f(\text{Re})$$

Danköehler suggested $\text{Re}^{1/2}$

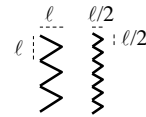
S_T for Flamelet Regime

- Other models with similar approach

$$(X.11) \quad \frac{S_T}{S_L} \cong 1 + \left(\frac{u'_{rms}}{S_L} \right)^n$$

with $n = n(u'_{rms}/S_L)$
e.g. $n > 1$ for small u'_{rms}/S_L
 $n < 1$ for large u'_{rms}/S_L

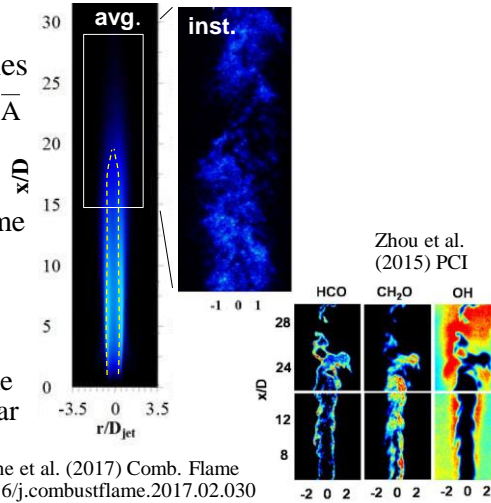
- compare reasonably well with experiments in flamelet regime (especially closer to wrinkled)
- e.g., SI engines: u'_{rms} scales with RPM (piston velocity) \Rightarrow can burn faster at higher RPM
- also, length scales not important



- Models fail as $S_L \rightarrow 0$ (e.g., due to flammability limits, quenching); predicts non-zero $S_T \rightarrow u'_{rms}$ same A

Example: S_T for Turbulent Jets

- Can measure S_T for using various approaches based on X.10 $S_T \equiv \dot{m} / \rho_u \bar{A}$
 - key is measuring average flame area
 - mean location in flame “brush”
 - center of average luminosity
 - maximum probability of flame location from planar laser imaging

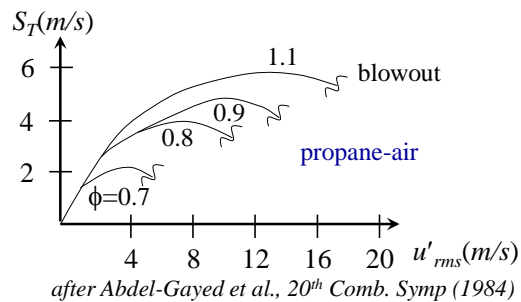


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Example: S_T for Turbulent Jets

- Turbulent flame speed initially increases with turbulence (u'_{rms})
- S_T also increases with S_L (ϕ here)
- S_T goes through maximum at high u'_{rms} / S_L
- **Why maximum?**
 - $S_L \neq \text{constant}$: stretch changes local laminar flamelets; high stretch can lead to extinction

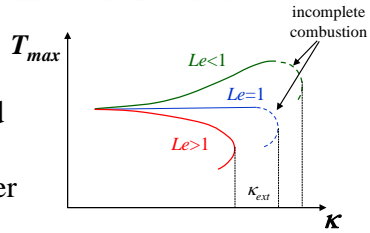
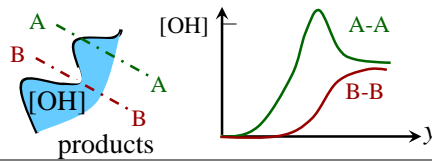


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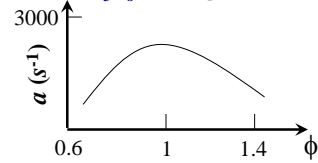
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Stretch and Flame Extinction

- Recall stretch can enhance or weaken laminar flames
- Turbulence enhances stretch (and scalar dissipation)
 - ⇒ produces higher gradients (higher diffusive fluxes, lower τ_{res})
 - ⇒ eventually extinguish flamelet
- Example evidence of local extinction measured with PLIF



Extinction Strain Rate for Laminar STP C_3H_8 -Air Stagn. Flame



reduced radical overshoot as approach extinction

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